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# Energy transmission through a double-wall structure with an acoustic enclosure: Rotational effect of mechanical links

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## Abstract

This paper investigates the rotational effect of mechanical links on energy transmission of a double-wall structure with an enclosure. A criterion is proposed to identify energy transmission mechanisms and predict the dominant transmitting path. Studies in different frequency ranges show a more significant energy transmission due to the rotational effect of the link at higher frequencies compared with lower ones. Comparison between the translational effect and the rotational effect on energy transmission shows that although both effects are important for the transmission mechanism analysis, the rotational effect on energy transmission is more remarkable at high frequencies for a soft translational link; whereas is insensitive for a stiff one. © 2005 Elsevier Ltd. All rights reserved.

*Keywords:* Energy transmission; Rotational effect; Double-wall structure; Links

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## 1. Introduction

Energy transmission through double-wall structures has been widely studied during the past decades [1,2]. A clear understanding of the energy transmission

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mechanism between the two panels is essential for vibroacoustic analysis or active noise control of such structures [3–10]. Most previous work focused on structures without any mechanical joints between the two walls, in which energy is entirely transmitted through the air gap, forming the acoustic transmitting path. Recently, studies on energy transmission between the two panels connected with mechanical links or connectors have been presented [11–16]. For example, Lin et al. [11] investigated the transmission of a plane sound wave through two infinite parallel plates connected by identical periodically spaced frames, and also compared the strengths of the two transmission paths (structural path through the studs and the airborne path through the cavity). Bao et al. [12] experimentally examined the effect of the mechanical path on sound transmission through double walls for active acoustic control applications. It was observed that, in some circumstances, a significant portion of energy could be transmitted from these links, referred to as structural transmitting path. In general, a realistic mechanical link conveys energy between the two panels through transverse forces and the moments, which are caused by the translational and rotational effects of the link, respectively. Effects of the translational link on energy transmission have been extensively investigated in [13–16]. In our previous work [13], the respective effects of the air gap and the links on energy transmission and noise insulation properties have been assessed. It was observed that the translational stiffness of the mechanical link and the aerostatic stiffness of the air gap are the two parameters governing the energy transmission process. A criterion was also proposed to analyze different transmission mechanisms and then to predict the dominant transmitting path.

As far as the rotational effect on sound/vibration transmission is concerned, there is a considerable amount of literature dealing with other configurations such as beams or plate-like structures [17–21]. For example, Petersson et al. and Sanderson et al. [17–19] investigated the important role played by the moment excitation in vibration transmission in built-up structure. Koh et al. [20] claimed that both force and moment excitations are critical in the energy transmission through beams or plate-like structures, in which moment excitations were considered through moment mobilities using Rayleigh–Ritz method. Gardonio et al. [21] studied translational and rotational excitations in the energy transmission of a vibrating mass isolated from a finite plate, and showed that at high frequencies, the energy was transmitted through axial and angular vibration simultaneously. Investigations by Goyder and White [22] revealed that, in the high frequency region, the vibrational power input to a beam or a plate due to a moment excitation exceeds that due to a force excitation. All these evidences suggest that the rotational effect is important to be considered in sound/vibration transmission analysis.

Compared with aforementioned configurations, double-plate structures with rotational links, however, received much less attention. One of a very few existing examples is the work of Takahashi [23], in which the sound radiation from periodically connected infinite double-panel structures were investigated. That work concluded that the stiffness of the connectors has important effects upon the radiated power. However, no systematic analysis has been reported so far to investigate the rotational effect of links on energy transmission for a double-wall structure coupled with

an acoustic enclosure. Criteria to identify the dominating transmitting mechanism are also lacking. In addition, the relative importance of the translational effect and the rotational effect on energy transmission also needs to be clarified in different frequency regions.

This paper attempts to answer these questions using a mechanically linked double-wall structure, radiating sound into a rectangular enclosure. A brief description of the modeling process is first presented. Numerical studies are then conducted to investigate the rotational effect on energy transmission and noise insulation properties. Based on an index defined as the ratio between the rotational stiffness of the link and the aerostatic stiffness of the air gap, a criterion for predicting the dominant transmission path is proposed. The relationship between the rotational effect and the translational effect on energy transmission is explored in different frequency regions. Finally, some conclusions are drawn.

## 2. Formulation

Consider a mechanically linked double-wall structure connected to an acoustic enclosure as shown in Fig. 1. The double-wall structure is composed of two homogeneous and isotropic rectangular panels a (upper) and b (lower), which are simply supported along their boundaries and separated by an air gap cavity with a volume  $V_g$  and a thickness  $h_g$ . A mechanical link is located at  $(x_m, y_m)$  to connect these two panels. Apart from the surfaces occupied by the two panels, all other surrounding walls of the air gap and the enclosure are acoustically rigid.

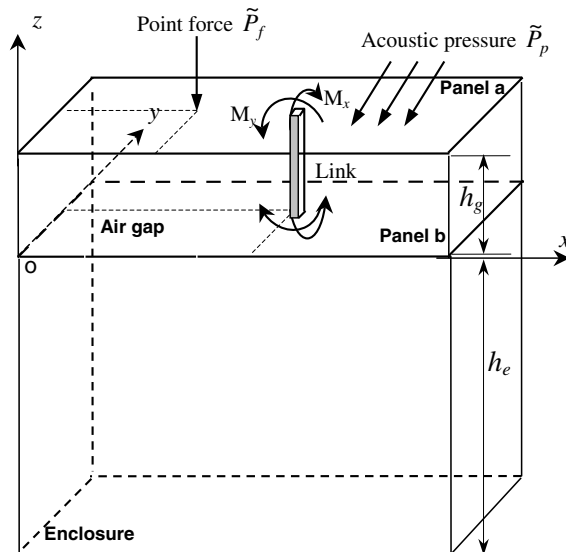


Fig. 1. A mechanically linked double-wall structure.

As the first step of energy transmission analysis, a vibroacoustic model, which includes the vibration of the two panels and the acoustic pressure inside the air gap and the enclosure, is established. For panel a, the equation of motion under an external excitation, which can either be a point force  $\tilde{P}_f$  at  $(x_f, y_f)$  or an acoustic pressure  $\tilde{P}_p$ , is expressed as

$$D_a \nabla^4 w_a + \rho_a h_a \frac{\partial^2 w_a}{\partial t^2} = \tilde{P}_f \delta(x - x_f, y - y_f) + \tilde{P}_p - f_m \cdot \delta(x - x_m, y - y_m) \\ - M_x \cdot \delta(x - x_m) \delta'(y - y_m) - M_y \cdot \delta'(x - x_m) \delta(y - y_m) \\ - P_g(z = h_g), \quad (1)$$

and for panel b,

$$D_b \nabla^4 w_b + \rho_b h_b \frac{\partial^2 w_b}{\partial t^2} = f_m \cdot \delta(x - x_m, y - y_m) + M_x \cdot \delta(x - x_m) \delta'(y - y_m) \\ + M_y \cdot \delta'(x - x_m) \delta(y - y_m) + P_g(z = 0) - P_e(z = 0), \quad (2)$$

where  $(w_a, D_a, \rho_a, h_a)$  and  $(w_b, D_b, \rho_b, h_b)$  are the transverse displacements (positive downwards), the flexible rigidities, the densities and the thicknesses of panels a and b, respectively.  $\delta$  is the Dirac delta function representing the concentrated source while  $\delta'$  is its derivation describing the moment.  $P_g$  and  $P_e$  are the acoustic pressures inside the air gap and the enclosure, respectively.  $f_m$ ,  $M_x$  and  $M_y$  are the force, the rotational moments about the  $x$ - and  $y$ -axis caused by the mechanical link, respectively. They can be simulated by a spring having a translational stiffness  $K_m$  as

$$f_m = K_m [w_a(x_m, y_m) - w_b(x_m, y_m)], \quad (3a)$$

and an equal rotational stiffness  $C_m$  in  $x$ - and  $y$ -directions as

$$M_x = C_m \theta_x, \quad M_y = C_m \theta_y. \quad (3b)$$

In Eq. (3b),  $\theta_x$  and  $\theta_y$  are the angular rotations described as

$$\theta_x = \frac{\partial w_a(x_m, y_m, t)}{\partial y} - \frac{\partial w_b(x_m, y_m, t)}{\partial y}, \\ \theta_y = - \left( \frac{\partial w_a(x_m, y_m, t)}{\partial x} - \frac{\partial w_b(x_m, y_m, t)}{\partial x} \right). \quad (4)$$

It is relevant to mention that the total pressure acting on the outer side of the panel a contains three parts, i.e. the incident pressure, the reflected pressure when the panel is assumed rigid and the radiated pressure due to vibration of the panel. The combination of the first two parts is usually referred to as the blocked pressure. It is generally accepted that, in air, the radiated pressure is rather low compared to the blocked pressure [10]. Therefore, the radiated pressure towards outside of the panel a is neglected in the present formulation.

Following the conventional modal superposition theory,  $w_a$  and  $w_b$  are decomposed over the mode shape functions  $\varphi_{a,ij}(x, y)$  and  $\varphi_{b,ij}(x, y)$  of panels a and b, respectively,

$$w_a(x, y, t) = \sum_{i=1}^M \sum_{j=1}^N \varphi_{a,ij}(x, y) q_{a,ij}(t), \tag{5a}$$

$$w_b(x, y, t) = \sum_{i=1}^M \sum_{j=1}^N \varphi_{b,ij}(x, y) q_{b,ij}(t), \tag{5b}$$

where  $q_{a,ij}(t)$  (or  $q_{b,ij}(t)$ ) is the  $ij$ -modal coordinate of panel a (or b). Using the orthogonality property of shape functions and considering the effect of viscous damping, Eqs. (1) and (2) can be expressed as

$$\begin{aligned} &\ddot{q}_{a,kl}(t) + 2\zeta_{a,kl}\omega_{a,kl}\dot{q}_{a,kl}(t) + \omega_{a,kl}^2q_{a,kl}(t) \\ &= \frac{1}{m_{a,kl}} \left\{ \int \int (\tilde{P}_p - P_g)\varphi_{a,kl} \, dx \, dy + \tilde{P}_f\varphi_{a,kl}(x_f, y_f) - f_m\varphi_{a,kl}(x_m, y_m) \right. \\ &\quad - \int \int M_x \cdot \delta(x - x_m)\delta'(y - y_m)\varphi_{a,kl} \, dx \, dy \\ &\quad \left. - \int \int M_y \cdot \delta'(x - x_m)\delta(y - y_m)\varphi_{a,kl} \, dx \, dy \right\}, \tag{6} \end{aligned}$$

$$\begin{aligned} &\ddot{q}_{b,kl}(t) + 2\zeta_{b,kl}\omega_{b,kl}\dot{q}_{b,kl}(t) + \omega_{b,kl}^2q_{b,kl}(t) \\ &= \frac{1}{m_{b,kl}} \left\{ \int \int (P_g - P_e)\varphi_{b,kl} \, dx \, dy + f_m\varphi_{b,kl}(x_m, y_m) \right. \\ &\quad + \int \int M_x \cdot \delta(x - x_m)\delta'(y - y_m)\varphi_{b,kl} \, dx \, dy \\ &\quad \left. + \int \int M_y \cdot \delta'(x - x_m)\delta(y - y_m)\varphi_{b,kl} \, dx \, dy \right\}, \tag{7} \end{aligned}$$

where  $\omega_{a,kl}$  (or  $\omega_{b,kl}$ ) and  $m_{a,kl}$  (or  $m_{b,kl}$ ) are the  $kl$ th natural angular frequency and the generalized mass of the  $kl$ th mode of panel a (or b), respectively,  $k = 1, \dots, M$ ;  $l = 1, \dots, N$ .

As far as the acoustic pressure  $P_g$  inside the gap cavity is concerned, one has the following classical wave equation and the associated constraint conditions based on velocity continuity:

$$\nabla^2 P_g - \frac{1}{c_o^2} \frac{\partial^2 P_g}{\partial t^2} = 0, \quad \frac{\partial P_g}{\partial \mathbf{n}} = \begin{cases} \rho \ddot{w}_a & \text{on panel a,} \\ -\rho \ddot{w}_b & \text{on panel b,} \\ 0 & \text{on the rigid wall,} \end{cases} \tag{8}$$

where  $\rho$  and  $c_o$  are the equilibrium fluid density and the sound velocity within the cavity, respectively.  $\mathbf{n}$  is the normal direction towards outside of the wall. Decomposing  $P_g$  on the basis of the acoustic mode shapes  $\psi_{g,j}$  as  $P_g = \sum_j \psi_{g,j} p_{g,j}(t)$  and applying Green's theorem [24], Eq. (8) can be rewritten as

$$\begin{aligned} & \ddot{p}_{g,j}(t) + 2\zeta_{g,j}\omega_{g,j}\dot{p}_{g,j}(t) + \omega_{g,j}^2 p_{g,j}(t) \\ &= \frac{\rho A c_o^2}{m_{g,jj} V_g} \left[ \sum_{k=1}^M \sum_{l=1}^N L_{j,kl}^{ag} \ddot{q}_{a,kl}(t) - \sum_{k=1}^M \sum_{l=1}^N L_{j,kl}^{bg} \ddot{q}_{b,kl}(t) \right], \quad j = 1, \dots, n_g, \end{aligned} \tag{9}$$

by introducing a modal loss factor  $\zeta_{g,j}$ . In Eq. (9),  $p_{g,j}(t)$ ,  $\omega_{g,j}$  and  $m_{g,jj}$  stand for the  $j$ th modal pressure amplitude, angular frequency and the generalized mass of the gap cavity, respectively.  $A$  is the area of the wall.  $L_{j,kl}^{ag}$  and  $L_{j,kl}^{bg}$  are the modal coupling coefficients. Similarly, the acoustic pressure  $P_e$  inside the enclosure satisfies

$$\ddot{p}_{e,j}(t) + 2\zeta_{e,j}\omega_{e,j}\dot{p}_{e,j}(t) + \omega_{e,j}^2 p_{e,j}(t) = \frac{\rho A c_o^2}{m_{e,jj} V_e} \sum_{k=1}^M \sum_{l=1}^N L_{j,kl}^{be} \ddot{q}_{b,kl}(t), \quad j = 1, \dots, n_e. \tag{10}$$

In Eq. (10), symbols with subscript “e” have the same meaning as those defined above but apply to the enclosure. In the case of harmonic excitation, Eqs. (6), (7), (9) and (10) can be combined into matrix form after derivations (see Appendix), and the displacement of each panel and the acoustic pressures inside the air gap and the enclosure can therefore be calculated. It is pertinent to mention that the derived equations are cast in a general form, considering the rotational and translational effects simultaneously. Should only rotational effect be considered, Eqs. (6) and (7) can be further simplified by setting  $K_m = 0$ .

Vibrations of the two panels are examined using the averaged quadratic velocity defined as

$$\langle V_a^2 \rangle = \frac{\omega^2}{2A} \int_A w_a w_a^* ds, \quad \langle V_b^2 \rangle = \frac{\omega^2}{2A} \int_A w_b w_b^* ds \tag{11}$$

for panels a and b, respectively. The asterisk denotes complex conjugate. Based on the fact that the area under the spectrum of  $\langle V^2 \rangle$  roughly represents the energy level within the analyzed frequency range, energy transmission between the two panels can be characterized by

$$\gamma_{pl} = \frac{\rho_b h_b \overline{\langle V_b^2 \rangle}}{\rho_a h_a \overline{\langle V_a^2 \rangle}}, \tag{12}$$

where  $\overline{\langle V_b^2 \rangle}$  and  $\overline{\langle V_a^2 \rangle}$  are, respectively, the areas under the spectra  $\langle V_b^2 \rangle$  and  $\langle V_a^2 \rangle$  defined as

$$\overline{\langle V_b^2 \rangle} = \int_{f_1}^{f_2} \langle V_b^2 \rangle df, \quad \overline{\langle V_a^2 \rangle} = \int_{f_1}^{f_2} \langle V_a^2 \rangle df, \tag{13}$$

where  $[f_1-f_2]$  is the frequency range of interest.

The acoustic insulation properties of the double-wall structure are then analyzed using the noise reduction index,  $\gamma_{NR}$ , defined as the difference between the sound pressure level at the top side of panel a and the averaged sound pressure level inside the enclosure as:

$$\gamma_{\text{NR}} = 10 \log \left( \frac{\langle P_{\text{out}}^2 \rangle}{P_{\text{ref}}^2} \right) - 10 \log \left( \frac{\langle P_{\text{e}}^2 \rangle}{P_{\text{ref}}^2} \right), \quad (14)$$

where  $\langle P_{\text{out}}^2 \rangle$  is the mean-square sound pressure averaged over the top side surface of panel a under an oblique incident plane wave excitation,

$$\langle P_{\text{out}}^2 \rangle = \frac{1}{2A} \int_A P_{\text{out}} P_{\text{out}}^* \, dS, \quad (15a)$$

$$\langle P_{\text{e}}^2 \rangle = \frac{1}{2V_{\text{e}}} \int_{V_{\text{e}}} P_{\text{e}} P_{\text{e}}^* \, dv, \quad p_{\text{ref}} = 20 \, \mu\text{Pa}. \quad (15b)$$

### 3. Results and discussions

Numerical simulations were performed to investigate the rotational effect on energy transmission between the two aluminum panels. Dimensions of the two panels are  $0.5 \times 0.35 \times 0.002 \, \text{m}^3$  for the upper panel a, and  $0.5 \times 0.35 \times 0.003 \, \text{m}^3$  for the lower one b (see Fig. 1). The depth of the enclosure is fixed at  $h_{\text{e}} = 0.55 \, \text{m}$ , while the depth of the air gap  $h_{\text{g}}$  is varied in different cases. A mechanical link, modeled as a spring with a translational stiffness  $K_{\text{m}}$  and a rotational stiffness  $C_{\text{m}}$ , is located at (0.3, 0.21) to connect the two panels. A harmonic excitation force with an amplitude of 1N is applied to panel a at (0.2, 0.14). Modal loss factors of 0.005 are assigned to both panels and 0.001 for the gap cavity and the enclosure. In the structural displacement and sound pressure decompositions, the number of modes used are (10,10) for the two panels, (9,7,2) for the gap cavity and (9,7,8) for the enclosure, which proved to be sufficient to properly cover the frequency range of interest.

#### 3.1. Effect of the rotational moment on energy transmission

By neglecting the translational effect of the link ( $K_{\text{m}} = 0$ ), the rotational effect on energy transmission is studied with  $C_{\text{m}} = 10^3 \, \text{N/rad}$  and  $C_{\text{m}} = 0$ . The averaged quadratic velocity of the two panels and the averaged sound pressure inside the cavities are plotted in Figs. 2(a) and (b), respectively. Comparing both cases shows a significant increase in the vibration energy transmission from panel a to b due to the rotational effect of the link (Fig. 2(a)). As a result, the averaged sound pressure inside the enclosure also increases. This increase is much more apparent at high frequencies, indicating an enhancing rotational effect on energy transmission as frequencies increase (Fig. 2(b)). Although the vibration level of panel a and the sound pressure level inside the air gap remain more or less the same before and after the consideration of  $C_{\text{m}}$ , resonance peaks are evidently altered, and some dominated by panel b can also be clearly identified in the spectra (denoted by a “•”), indicating a strong coupling between the two panels and a significant energy transmission due to the rotational effect of the link.

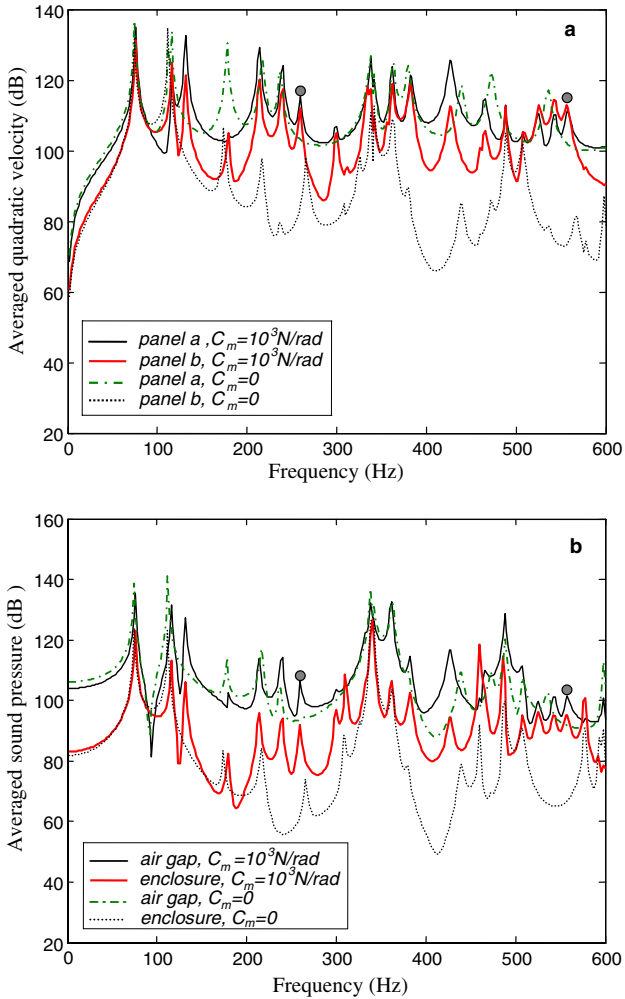


Fig. 2. Energy transmission without considering translational effect of the mechanical link when  $h_g/h_e = 0.2$ : (a). Averaged quadratic velocity of panels a and b; and (b). Averaged sound pressure level inside the air gap and the enclosure.

As reported in [13], the translational effect of mechanical links changes energy transmission mechanism, in which a criterion is presented to predict the dominant transmitting path. That criterion is revisited here by considering the rotational effect. Fig. 3(a) illustrates a tendency plot of  $\gamma_{pl}$  for different rotational stiffness with respect to  $h_g/h_e$ . Given a  $C_m$ ,  $\gamma_{pl}$  first undergoes a subtle drop, then enters into a smooth drop region before finally reaching a plateau with the increase of  $h_g/h_e$ . The occurrence of each zone depends on the stiffness  $C_m$ . A stiffer link apparently pushes these zones at lower  $h_g/h_e$  values. All curves exhibit the following characteristics: In the first zone,  $\gamma_{pl}$  is extremely sensitive to  $h_g/h_e$ , suggesting that a slight increase in  $h_g/h_e$  can result



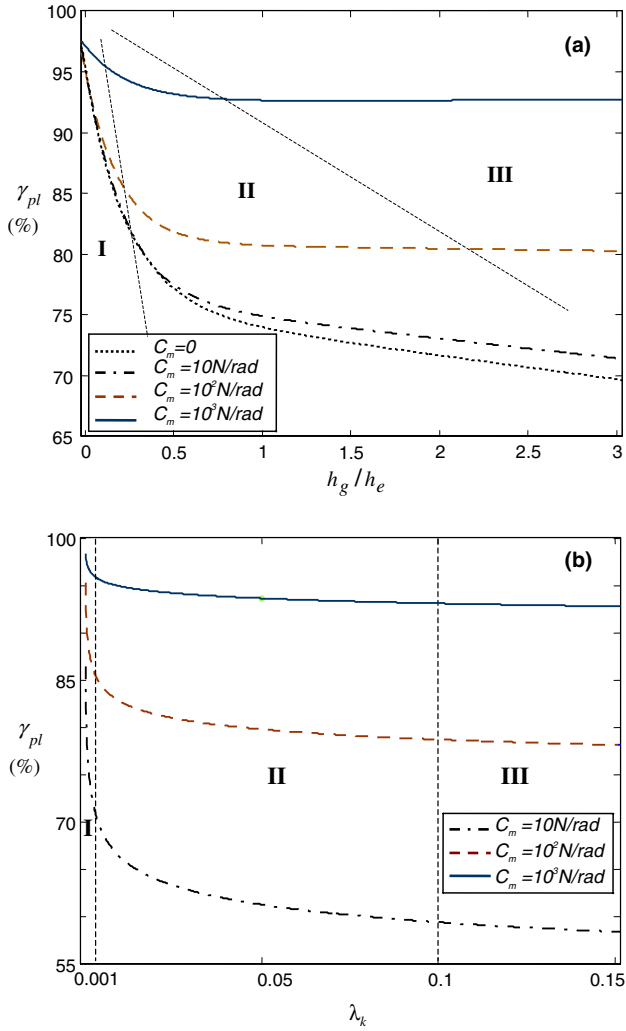


Fig. 3. Tendency plot of  $\gamma_{pl}$  showing the energy transmission between two walls: (a) with respect to  $h_g/h_e$ ; and (b) with respect to  $\lambda_k$ .

in an obvious attenuation of the energy transmission between the two panels. This phenomenon is most remarkable with a soft link, with most of energy being transmitted through the air gap rather than the mechanical link. In the second zone, the gap effect is weakened and energy is transmitted through the link and the air gap simultaneously. In the third zone,  $\gamma_{pl}$  is insensitive to  $h_g/h_e$ , implying that the mechanical link dominates the energy transmission path.

Another observation is that, with an increasing  $C_m$ ,  $\gamma_{pl}$  increases within a shortened first two zones and an elongated plateau zone, showing a remarkable rotational effect on energy transmission. Obviously,  $C_m$  and  $h_g$  are the two critical parameters

to separate these zones. Their effects on energy transmission can be better shown by defining a normalized parameter as

$$\lambda_k = \frac{C_m}{K_g}, \quad (16)$$

where  $K_g$  is the aerostatic stiffness of the gap cavity [13]

$$K_g \approx \rho c_o^2 A^2 (L_{o,11}^{bg})^2 / V_g, \quad (17)$$

in which  $L_{o,11}^{bg}$  is the modal coupling coefficient between the cavity mode (0,0,0) of the air gap and the structural mode (1,1) of the panel b.

The tendency plot of  $\gamma_{pl}$  with respect to the normalized parameter  $\lambda_k$  is given in Fig. 3(b). Evidently, three zones can be systematically demarcated with rough delimitations at  $\lambda_k = 0.001$  and 0.1. That is, when  $\lambda_k \in (0, 0.001]$ , energy will be mainly transmitted from the acoustic path; while when  $\lambda_k \in [0.1, \infty)$ , the structural transmission is dominant; when  $\lambda_k \in [0.001, 0.1]$ , energy is conveyed through both the acoustic and structural paths simultaneously. Although the demarcation lines are vague, this criterion can be used as a simple tool to roughly predict the main energy transmission path.

### 3.2. Noise reduction

Effects of transmission paths on noise reduction of the double-wall structure are investigated. An oblique plane wave having an amplitude of 1 Pa, an elevation angle of  $60^\circ$  and an azimuth angle of  $30^\circ$  is exerted on panel a. Fig. 4(a) illustrates the noise reduction index  $\gamma_{NR}$  when  $h_g/h_e = 0.2$  for three different cases:  $C_m = 0, 25$  and  $10^3$  N/rad. Comparison between the curve for  $C_m = 25$  N/rad with that for  $C_m = 0$  shows that both curves almost coincide with each other. The reason is that the normalized parameter  $\lambda_k (= 6.8 \times 10^{-4})$  falls into zone I, in which energy is mostly transmitted through acoustic path. Such a soft link, therefore, has negligible effect on noise reduction. When  $C_m$  is increased to  $10^3$  N/rad (corresponding to  $\lambda_k = 2.7 \times 10^{-2}$ ), however, except for the *first two* modes of the coupled system dominated by the two panels, a clear decrease in  $\gamma_{NR}$  is observed, meaning that most energy is transmitted through rotational moments, resulting in a much reduced noise insulation.

Fig. 4(b) shows the variation of  $\gamma_{NR}$  when  $C_m = 10^2$  N/rad with: (1)  $h_g/h_e = 0.05$  (in vacuo); (2)  $h_g/h_e = 0.05$  ( $\lambda_k = 6.8 \times 10^{-4}$ ) and (3)  $h_g/h_e = 0.25$  ( $\lambda_k = 3.4 \times 10^{-3}$ ). The obvious difference between the two cases with shallow gaps ( $h_g/h_e = 0.05$ ) with and without air indicates a significant energy transmission through the acoustic path. When  $h_g/h_e$  is increased to 0.25,  $\gamma_{NR}$  curve approaches the in vacuo one. This is understandable because the increase in the gap depth weakens acoustic transmitting path, accompanied by a more dominant energy transmission through the link.

### 3.3. Rotational effect vs. frequencies

Rotational effect on energy transmission in different frequency regions is investigated. Two representative frequency bands are chosen for this purpose: a lower

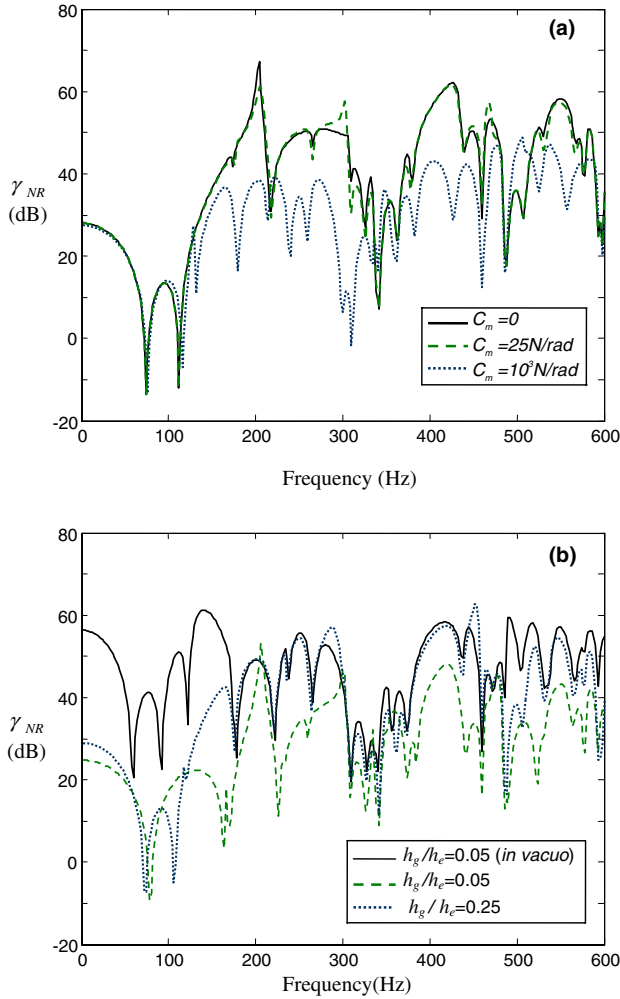


Fig. 4. Noise reduction index  $\gamma_{NR}$  ( $h_g/h_e = 0.2$ ): (a) Rotational effect of the mechanical link; and (b) Effect of the depth of the gap.

band covering 0 to 200 Hz, and a higher band ranging from 400 to 600 Hz. Using three different gap depths  $h_g/h_e = 0.1, 0.2$  and  $0.5$ , Figs. 5(a) and 5(b) depict the variations of  $\gamma_{pl}$  with respect to  $\lambda_k$  for the higher frequency band and the lower one, respectively. For both frequency bands,  $\gamma_{pl}$  increases with the increase of  $\lambda_k$ , indicating an enhanced energy transmission through rotational moments when the link is getting stiffer. A comparison between Figs. 5(a) and 5(b) clearly shows that the increase rate of  $\gamma_{pl}$  at high frequencies is much higher than that at low frequencies. Taking the case of  $h_g/h_e = 0.5$  (solid line,  $K_g \approx 1.5 \times 10^4 \text{ N/m}$ ) as an example, a change in  $\lambda_k$  between  $10^{-5}$  and  $10^{-0.5}$  leads to 26.1% increase in  $\gamma_{pl}$  in the higher frequency band (Fig. 5(a)), while only 8.9% in the lower one

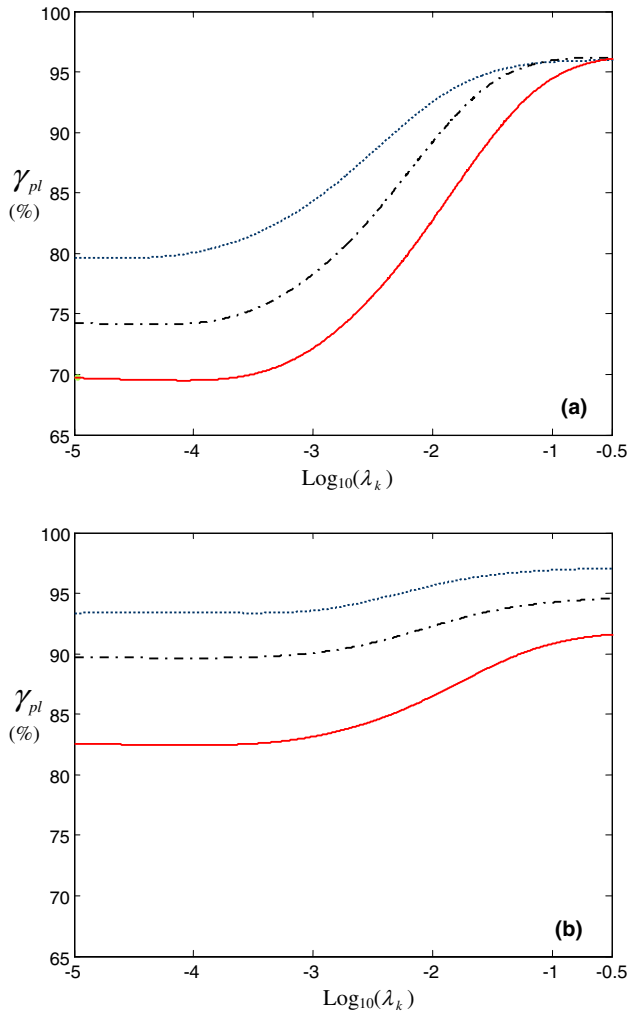


Fig. 5. Energy transmission between two walls for different  $h_g/h_e$  in the frequency region: (a) [400–600] Hz; and (b) [0–200] Hz.  $h_g/h_e = 0.1$ :  $\cdots$ ;  $h_g/h_e = 0.2$ :  $-\cdot-$ ;  $h_g/h_e = 0.5$ :  $—$ .

(Fig. 5(b)). This observation implies a stronger rotational effect on energy transmission at high frequencies compared with lower ones. This is consistent with previous observations made on simple structures such as infinite beams or plates, whose rotational mobility increases with frequencies, whilst translational mobility usually keeps unchanged [13]. Both Figs. 5(a) and 5(b) also show that the variation of  $\gamma_{pl}$  is narrowed down when  $h_g/h_e$  is getting smaller. This is quite reasonable because, for a shallower gap, acoustic transmission becomes more dominant, such reducing the effect of the link.

3.4. Rotational effect vs. translational effect

Practical mechanical links provide energy transmission path through both translation and rotation simultaneously. An example is given in Fig. 6, when  $h_g/h_c = 0.2$  ( $K_g \approx 3.6 \times 10^4$  N/m), to show the effects of  $K_m$  and  $C_m$  on energy transmission. Different areas can be identified in which energy transmission process takes place through different paths. Given low  $K_m$  values,  $\gamma_{pl}$  moves from area I through II to III as  $C_m$  increases, basically following the same tendency as described in Section 3.1. Keeping  $C_m$  constant at small values and varying  $K_m$  result in a similar variation in  $\gamma_{pl}$  (areas I, IV and V), which is consistent with previous investigations [13]. Within these two strips, energy is transmitted either completely through the air gap (area I), or together with rotation (II and III) or translation (IV and V) of the link. The two strips denoted by III and V represent areas dominated by a strong rotational and translational link, respectively. Their intersection area VI corresponds to a very hard link in both rotation and translation. Note that the variation in  $\gamma_{pl}$  is rather smooth in these areas, the links can be regarded as rigid compared with the two panels. Area VII forms a complex region in which energy is transmitted through a combination of all possible paths (air gap, rotational and translational links).

In the presence of the translational effect, rotational effect on energy transmission in different frequency regions is revisited. Fig. 7 depicts the variations of  $\gamma_{pl}$  in the higher and lower frequency bands, when  $K_m = 10^3, 3 \times 10^4$  and  $10^6$  N/m. It can be seen that the general tendency of the curves is similar to Fig. 5. A pronounced rotational effect occurs in the higher frequency band, especially for the structure with a soft translational link, e.g.,  $K_m = 10^3$  N/m. With the increase of  $K_m$ , however, the variation of  $\gamma_{pl}$  significantly reduces, from 20% for  $K_m = 10^3$  N/m (solid line) to 3% for  $K_m = 10^6$  N/m (dotted line), indicating a much weakened rotational effect for stiff translational links.

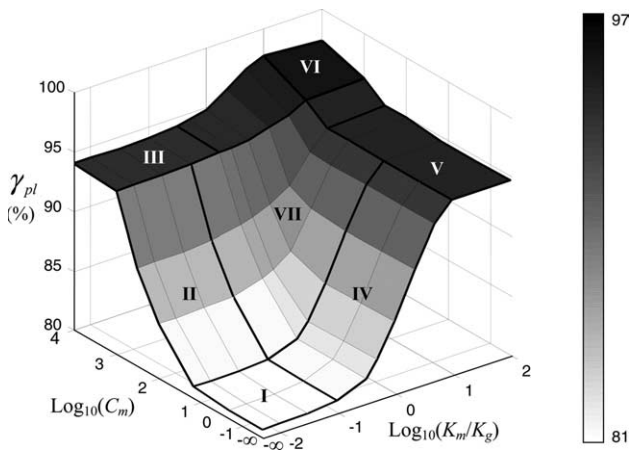


Fig. 6. Relationship between the rotational effect and the translational effect on energy transmission ( $h_g/h_c = 0.2$ ).

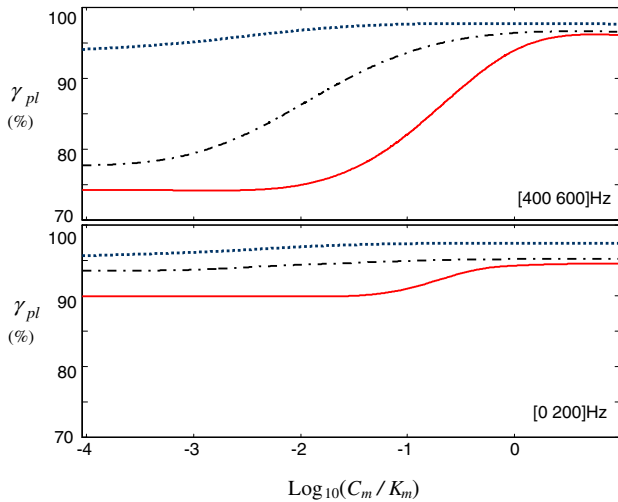


Fig. 7. Energy transmission between two walls ( $h_g/h_e = 0.2$ ) for different  $K_m$  in the frequency region: (a) [400–600] Hz; and (b) [0–200] Hz.  $K_m = 3 \times 10^4$  N/m: —;  $K_m = 2 \times 10^5$  N/m: - - -;  $K_m = 10^6$  N/m: ···.

#### 4. Conclusions

Energy transmission through rotational effect of mechanical links in a double wall structure coupled with an acoustic enclosure was investigated in this paper. Based on a fully coupled vibroacoustic model, numerical simulations were performed with a view to establish a simple criterion for determining the most dominant path for energy transmission. The following conclusions can be drawn.

- (1) The rotational effect of mechanical links on sound/vibration transmission of double-wall structures is obvious. It not only enhances the coupling between the two panels, but also intensifies the energy exchange between them, leading to an increased sound field in the enclosure, especially at higher frequencies.
- (2) The rotational stiffness of the links and the aerostatic stiffness of the air gap are found to be the key parameters governing the energy transmission process. The ratio between the two stiffness terms, i.e.,  $\lambda_k$ , can be used to roughly predict the dominant transmission path (acoustic or structural). Three zones are shown to exist in which different energy transmission mechanisms are involved. Energy is mainly transmitted from the acoustic path when  $\lambda_k \in (0, 0.001]$ , from the structural path when  $\lambda_k \in [0.1, \infty)$  or from both when  $\lambda_k \in [0.001, 0.1]$ .
- (3) The rotational effect exhibits much stronger influences on energy transmission at high frequencies due to the increasing mobility of the panels. This effect is further amplified when the double-wall structure has a soft translational link.

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**Appendix**

In the case of harmonic excitation, Eqs. (6), (7), (9) and (10) can be combined into matrix form as

$$\begin{bmatrix} H_{11} & H_{12} & H_{13} & 0 \\ H_{21} & H_{22} & H_{23} & H_{24} \\ H_{31} & H_{32} & H_{33} & 0 \\ 0 & H_{42} & 0 & H_{44} \end{bmatrix} \begin{Bmatrix} \mathbf{A} \\ \mathbf{B} \\ \mathbf{C} \\ \mathbf{D} \end{Bmatrix} = \begin{Bmatrix} \mathbf{F}_a \\ \mathbf{0} \\ \mathbf{0} \\ \mathbf{0} \end{Bmatrix}.$$

where

$$H_{11} = M_a + K_m \Phi^T(x_m, y_m) \Phi(x_m, y_m) + C_m [\Psi_x^T(x_m, y_m) \Psi_x(x_m, y_m) - \Psi_y^T(x_m, y_m) \Psi_y(x_m, y_m)],$$

$\Phi(x_m, y_m) = [\varphi_{11}(x_m, y_m), \varphi_{12}(x_m, y_m), \dots, \varphi_{MN}(x_m, y_m)]$ , ( $\varphi_{ij} = \varphi_a, ij = \varphi_b, ij$  for the two panels with same boundary condition)

$$\Psi_x(x_m, y_m) = \left[ \left. \frac{\partial \varphi_{11}(x, y)}{\partial x} \right|_{x=x_m, y=y_m}, \dots, \left. \frac{\partial \varphi_{MN}(x, y)}{\partial x} \right|_{x=x_m, y=y_m} \right],$$

$$\Psi_y(x_m, y_m) = \left[ \left. \frac{\partial \varphi_{11}(x, y)}{\partial y} \right|_{x=x_m, y=y_m}, \dots, \left. \frac{\partial \varphi_{MN}(x, y)}{\partial y} \right|_{x=x_m, y=y_m} \right],$$

$$H_{12} = -K_m \Phi^T(x_m, y_m) \Phi(x_m, y_m) - C_m [\Psi_x^T(x_m, y_m) \Psi_x(x_m, y_m) - \Psi_y^T(x_m, y_m) \Psi_y(x_m, y_m)],$$

$$H_{22} = M_b + K_m \Phi^T(x_m, y_m) \Phi(x_m, y_m) + C_m [\Psi_x^T(x_m, y_m) \Psi_x(x_m, y_m) - \Psi_y^T(x_m, y_m) \Psi_y(x_m, y_m)].$$

The expressions of other terms, e.g.,  $M_a$ ,  $M_b$ ,  $H_{13}$ , ... and  $H_{44}$ , are the same as those presented in our previous work. The readers can refer to [13] for details.

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