Subject Description Form

Subject Code	AAE4107				
Subject Title	Aircraft Gas Turbine Engine Systems				
Credit Value	3				
Level	4				
Pre-requisite/ Co-requisite/ Exclusion	Pre-requisite: AAE2102/IC2133 Aircraft Manufacturing and Maintenance Fundamentals				
Objectives	To provide students with knowledge of aircraft gas turbine engine systems and application in engine monitoring and maintenance				
Intended Learning	Upon completion of the subject, students will be able to:				
Outcomes	a. Acquire good understanding of aircraft turbine engine design and construction.				
	b. Demonstrate good understanding of compressor stall/surge and its prevention.				
	c. Apply their knowledge and skills to explain the limitations of aircraft gas turbine engines under normal and abnormal operational conditions.				
Subject Synopsis/ Indicative	Basic Aircraft Turbine Engine Design and Construction				
Syllabus	Constructional arrangement and operation of turbojet, turbofan, turboshaft, turboprop.				
	Compressor stall/surge				
	Causes and effects of compressor stall and surge and its prevention. Bearings and Seal				
	Constructional features and principles of operation.				
	Lubricants and Fuel				
	Properties and specifications; Fuel additives; Safety precautions.				
	Lubrication Systems				
	System operation/lay-out and components.				
	Fuel Systems				
	Operation of engine control and fuel metering systems including electronic engine control (FADEC); systems lay-out and components.				

Air Systems

Operation of engine air distribution and anti-ice control systems, including internal cooling, sealing and external air services.

Starting and Ignition Systems

Operation of engine start systems and components; ignition systems and components; maintenance safety requirements

Engine Indication Systems

Exhaust gas temperature / interstage turbine temperature; engine thrust indication: engine pressure ratio, engine turbine discharge pressure or jet pipe pressure systems; oil pressure and temperature; fuel pressure and flow; engine speed; vibration measurement and indication; torque; power.

Power Augmentation Systems

Operation and applications; water injection, water methanol; afterburner systems.

Turbo-prop Engines

Gas coupled/free turbine and gear coupled turbines; reduction gears; integrated engine and propeller controls; overspeed safety devices.

Turbo-shaft engines

Arrangements, drive systems, reduction gearing, couplings, control systems.

Auxiliary power units (APUs)

Purpose, operation, protective systems.

Powerplant Installation

Configuration of firewalls, cowlings, acoustic panels, engine mounts, antivibration mounts, hoses, pipes, feeders, connectors, wiring looms, control cables and rods, lifting points and drains.

Engine Monitoring and Ground Operation

Procedures for starting and ground run-up; interpretation of engine power output and parameters; trend (including oil analysis, vibration and boroscope) monitoring; inspection of engine and components to criteria, tolerances and data specified by engine manufacturer; compressor washing/cleaning; foreign object damage.

Engine Storage and Preservation

Preservation and depreservation for the engine and accessories / systems.

Teaching/Learnin g Methodology	Lectures are used to deliver the fundamental knowledge in relation to aircraft ga turbine engines (outcomes a to c). Tutorials are used to illustrate the applications of fundamental knowledge of practical situations (outcomes a to c).						
	Teaching/Learning Methodology	Intended su covered	bject learning outcomes to be				
		a	1	0	с		
	1. Lecture	~	v	/	✓		
	2. Tutorial	~	v	(~		
Assessment Methods in Alignment with Intended Learning	Specific assessment methods/tasks	% weighting		ided subject learning omes to be assessed			
Outcomes			а	b	с		
	1. Assignments / Quizzes	50%	~	~	✓		
	2. Final examination	50%	~	~	~		
	Total	100 %					
	 Explanation of the appropriateness of the assessment methods in assessing the intended learning outcomes: Overall Assessment: 0.5 × End of Subject Examination + 0.5 × Continuous Assessment Examination is adopted to assess students on the overall understanding and the ability of applying the concepts. It is supplemented by continuous assessment including assignments and closed-book quizzes. The continuous assessment is aimed at enhancing the students' comprehension and assimilation of various topics of the syllabus. 						
Student Study Effort Expected	Class contact:						
p	 Lectures 				36 Hrs.		
	Tutorials				3 H	rs.	
	Other student study effort:						
	 Assignments 	 Assignments 					
	 Self-study 	 Self-study 				rs.	
	Total student study effort				105 Hı	rs.	

Reading List and References	1.	EASA Module 15 Gas Turbine Engine, Aircraft Technical Book Co. 4 th Edition
	2.	The Jet Engine, Rolls Royce, Latest Edition
	3.	Mattingly, J.D., Boyer, K.M., von Ohain, H., Elements of Propulsion: Gas Turbines and Rockets, AIAA, 2016.
	4.	Aircraft Powerplants, Bent & McKinley, McGraw-Hill, 4th Edition
	5.	Aircraft Gas Turbine engine Technology, Irwin E Tregar, McGraw-Hill, 2 nd Edition
	6.	Thrust for flight, Thomson, W. (William), Longman, 2 nd Edition
	7.	Aircraft powerplants., Kroes, Michael J.; Thomas W. Wild, McGraw-Hill, Ninth Edition.
	8.	Aero engine combustor casing : experimental design and fatigue studies, Panigrahi, Shashi Kanta; Niranjan Sarangi, Boca Raton, 2017
	9.	Axial Turbine Aerodynamics for Aero-Engines: Flow Analysis and Aerodynamics Design, Zou, Zhengping ; Wang, Songtao ; Liu, Huoxing ; Zhang, Weihao, Springer Singapore, 2018

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