

## Subject Description Form

<b>Subject Code</b>	AAE4107
<b>Subject Title</b>	Aircraft Gas Turbine Engine Systems
<b>Credit Value</b>	3
<b>Level</b>	4
<b>Pre-requisite/ Co-requisite/ Exclusion</b>	<b>Pre-requisite:</b> AAE3003 Aircraft Propulsion Systems <b>AND</b> AAE2102/IC2133 Aircraft Manufacturing and Maintenance Fundamentals
<b>Objectives</b>	To provide students with knowledge of aircraft gas turbine engine systems and application in engine monitoring and maintenance
<b>Intended Learning Outcomes</b>	Upon completion of the subject, students will be able to: <ul style="list-style-type: none"> <li>a. Acquire good understanding of aircraft turbine engine design and construction; and</li> <li>b. Demonstrate good understanding of compressor stall/surge and its prevention; and</li> <li>c. Apply their knowledge and skills to explain the limitations of aircraft gas turbine engines under normal and abnormal operational conditions.</li> </ul>
<b>Subject Synopsis/ Indicative Syllabus</b>	<p><b>Inlet</b> - Compressor inlet ducts; Effects of various inlet configurations; Ice protection.</p> <p><b>Compressors</b> - Axial and centrifugal types; Constructional features and operating principles and applications; Wide-chord fan technology; Fan balancing; Operation: Causes and effects of compressor stall and surge; Methods of air flow control: bleed valves, variable inlet guide vanes, variable stator vanes, rotating stator blades; Compression ratio.</p> <p><b>Combustion Section</b> - Different designs of combustion chambers; Constructional features and principles of operation.</p> <p><b>Turbine Section</b> - Operation and characteristics of different turbine blade types; Blade to disk attachment; Nozzle guide vanes; Causes and effects of turbine blade stress and creep; Turbine cooling</p> <p><b>Exhaust</b> - Constructional features and principles of operation; Convergent, divergent and variable area nozzles; Engine noise reduction; Thrust reversers.</p>
<b>Teaching/Learning Methodology</b>	<p>Lectures are used to deliver the fundamental knowledge in relation to aircraft gas turbine engines (outcomes a to c).</p> <p>Tutorials are used to illustrate the applications of fundamental knowledge to practical situations (outcomes a to c).</p>

	Teaching/Learning Methodology	Intended subject learning outcomes to be covered			
		a	b	c	
	1. Lecture	✓	✓	✓	
	2. Tutorial	✓	✓	✓	
<b>Assessment Methods in Alignment with Intended Learning Outcomes</b>	Specific assessment methods/tasks	% weighting	Intended subject learning outcomes to be assessed		
			a	b	c
	1. Assignments / Quizzes	50%	✓	✓	✓
	2. Final examination	50%	✓	✓	✓
	Total	100 %			
<p>Explanation of the appropriateness of the assessment methods in assessing the intended learning outcomes:</p> <p>Overall Assessment:</p> <p><math>0.5 \times \text{End of Subject Examination} + 0.5 \times \text{Continuous Assessment}</math></p> <p>Examination is adopted to assess students on the overall understanding and the ability of applying the concepts. It is supplemented by continuous assessment including assignments and closed-book quizzes. The continuous assessment is aimed at enhancing the students' comprehension and assimilation of various topics of the syllabus.</p>					
<b>Student Study Effort Expected</b>	Class contact:				
	▪ Lectures		36 Hrs.		
	▪ Tutorials		3 Hrs.		
	Other student study effort:				
	▪ Assignments		20 Hrs.		
	▪ Self-study		46 Hrs.		
Total student study effort		<b>105 Hrs.</b>			
<b>Reading List and References</b>	<ol style="list-style-type: none"> <li>EASA Module 15 Gas Turbine Engine, Aircraft Technical Book Co. 4<sup>th</sup> Edition</li> <li>The Jet Engine, Rolls Royce, Latest Edition</li> <li>Mattingly, J.D., Boyer, K.M., von Ohain, H., Elements of Propulsion: Gas</li> </ol>				

	Turbines and Rockets, AIAA, 2016.
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July 2021